

5

# Cryogenics for the MuCool Test Area (MTA)

C. Darve, B. Norris, L. Pei

Fermi National Accelerator Laboratory Batavia, IL, 60500, USA

COOL'05 – Galena, IL

### Headlines

Scope

**Helium Facility** 

Hydrogen Facility

**Applications** 

- 1) Past: Convection LH<sub>2</sub> absorber
- 2) Present: SC Solenoid magnet
- 3) Future: Forced-flow LH<sub>2</sub> absorber, (Convection run 2)

# The MTA is a Fermilab test facility constructed to support new developments in the physics and engineering of muon beam cooling MTA cryo-engineering requirements

- > Design, prototype and bench test cooling-channel components
- Provide helium and nitrogen refrigeration
- Provide liquid hydrogen

 Install equipment, instrumentation, DAQ & control in compliance with Fermilab safety policy (ES&H)

Previous US H<sub>2</sub> experiments : E158, SAMPLE, G0, Fermilab hydrogen experts

### Important reference and guideline

FERMILAB: "Guidelines for the Design, Fabrication, Testing, Installation and Operation of LH<sub>2</sub> Targets – 20 May 1997" by Del Allspach et al. Fermilab RD\_ESH\_010– 20 May 1997

NASA: "SAFETY STANDARD FOR HYDROGEN AND HYDROGEN SYSTEMS: Guidelines for Hydrogen System Design, Materials Selection, Operations, Storage, and Transportation"



### Applications

- LH<sub>2</sub> Absorbers:
  - Convection scheme (<100W?) MICE design
  - Forced-flow scheme (up to 300W) MuCool design
- S.C. solenoid magnet for RF cavity test (long term: LH<sub>2</sub> absorber)
- Other applications using HP gaseous hydrogen : Muon Inc.

# Beam dump BEAM DIRECTION

### Proposed cooling channel test bench



# **Ionization & Cryogenic Materials**

Material	Z	A	$\langle Z/A \rangle$	Nuclear <sup>a</sup> collision length $\lambda_T$ $\{g/cm^2\}$	$\begin{array}{l} \text{Nuclear} \ ^a \\ \text{interaction} \\ \text{length} \ \lambda_I \\ \{ \text{g/cm}^2 \} \end{array}$	$\frac{dE/dx _{\min}}{\left\{\frac{MeV}{g/cm^2}\right\}}$	<sup>b</sup> Radiat {g/cm <sup>2</sup>	ion length $X_0$ $^2$ {cm}	c Density $\{g/cm^3\}$ $(\{g/\ell\}$ for gas)	Liquid boiling point at 1 atm(K)	$\begin{array}{c} \text{Refractive} \\ \text{index } n \\ ((n-1) \times 10^6 \\ \text{for gas}) \end{array}$
H <sub>2</sub> gas	1	1.00794	0.99212	43.3	50.8	(4.103)	61.28 d	(731000)	(0.0838)[0.0899]		[189.2]
H <sub>2</sub> liquid	1	1.00794	0.99212	43.3	50.8	4.034	61.28 <sup>d</sup>	866	0.0708	20.39	1.112
$D_2$	1	2.0140	0.49652	45.7	54.7	(2.052)	122.4	724	0.169[0.179]	23.65	1.128[138]
He	2	4.002602	0.49968	49.9	65.1	(1.937)	94.32	756	0.1249[0.1786]	4.224	1.024 [34.9]
Li	3	6.941	0.43221	54.6	73.4	1.639	82.76	155	0.534		
Be	4	9.012182	0.44384	55.8	75.2	1.594	65.19	35.28	1.848		_
C	6	12.011	0.49954	60.2	86.3	1.745	42.70	18.8	2.265 #		
N <sub>2</sub>	7	14.00674	0.49976	61.4	87.8	(1.825)	37.99	47.1	0.8078[1.250]	77.36	1.205 [298]
0 <sub>2</sub>	8	15,9994	0.50002	63.2	91.0	(1.801)	34.24	30.0	1.141[1.428]	90.18	1.22 [296]
$F_2$	9	18.9984032	0.47372	65.5	95.3	(1.675)	32.93	21.85	1.507[1.696]	85.24	[195]
Ne	10	20.1797	0.49555	66.1	96.6	(1.724)	28.94	24.0	1.204/0.90051	27.09	1.092 [67.1]
AL	13	26.981539	0.48181	70.6	106.4	1.615	24.01	8.9	2.70	-126-13	

.....

----



Ionization coolin

" 3ethe-Bloch multiple scatterin

### **Cryogenic Properties of Various Gasses**

M. Green: "Hydrogen Safety Issues compared to Safety Issues with Methane and Propane", CEC'05 Parameter He N<sub>2</sub> H<sub>2</sub> CH<sub>4</sub> C<sub>3</sub>H<sub>8</sub>

1 al allicici		1.12	2	011 4	0 3 8
Triple point temperature T <sub>t</sub> (K)	2.177^	63.15	13.81*	90.69	91.46
Heat of fusion @ $T_t (J g^{-1})$	-NA-	25.3	59.5*	58.41	<b>79.9</b> 7
Boiling temp. T b @ 1 bar (K)	4.222	77.35	20.28*	111.67	230.46
$\label{eq:liquid} Liquid \ density \qquad \rho_{\scriptscriptstyle 1} \ @ \ T_{\scriptscriptstyle b} \ (kg \ m^{-3})$	124.9	<b>807</b>	70.8*	422.4	585.3
Gas density $\rho_g @ T_b (kg m^{-3})$	16.89	4.622	1.339*	1.816	2.497
Gas to liquid volume ratio at T b	7.395	175.6	52.87*	232.6	298.0
Gas V 293 to Liquid V Ratio	699.4	645.6	792.9	591.4	379.0
Gas C $_{\rm p}$ @ T $_{\rm b}$ (J g $^{-1}$ K $^{-1}$ )	9.144	1.341	12.24*	2.218	1.642
Heat of vaporization @ T $_{b}$ (J g $^{-1}$ )	20.7	198.8	445*	510.8	424.8
Heat flux for $\Delta T=300$ -T <sub>b</sub> (kWm <sup>-2</sup> )	~200	~27	~93	~47	-NA-
Broken vacuum heat flux (kWm <sup>-2</sup> )	~35	~1.6	~19	~0.31	-NA-
Critical temperature $T_{c}(K)$	5.195	126.2	32.98	190.6	368.8
Critical pressure P <sub>c</sub> (MPa)	0.228	3.39	1.29	4.59	4.36

^ The lambda point temperature for helium. For helium liquid, gas, and solid can't coexist.
\* This data is for para hydrogen. Ortho hydrogen changes to para hydrogen at < 100 K.</li>

CEC 30 August 2005

# **Ionization & Cryogenic Materials**

### Properties of gaseous (normal) hydrogen are as follows:

Reference temperature	68°F	527.7° R	293.1° K
Standard pressure (1 atm) psia	14.69 kPa	101.325 abs	
Density (at 527.7° R & 1 atm)	.00523 lb/ft <sup>3</sup>	83.7 g/m <sup>3</sup>	5.52
Specific Volume (at 527.7° R & 1 atm)	191.4 ft <sup>3</sup> /lb	0.0119 m <sup>3</sup> /g	
Specific Heat	Cp= 3.425 Btu/lb-R Cv= 2.419 Btu/lb-R	Cp= 14.33 J/g-k Cv= 10.12 J/g-k	
Velocity of Sound	4246 ft/sec	1294 m/sec	
Heat of Combustion	Low = 51596 Btu/lb High = 61031Btu/lb	Low= 119.93 kJ/g High= 141.86 kJ/g	
Flammability limits Hydrogen-air mixture Hydrogen-oxygen mixture	Lower= 4.0 % volume Lower= 4.0 % volume	Upper= 75 % volume Upper= 95 % volume	
Explosive limits Hydrogen-air mixture Hydrogen-oxygen mixture	Lower= 18.3 % volume Lower= 15.0 % volume	Upper= 59 % volume Upper= 90 % volume	
Minimum spark ignition energy at 1 atm			
In air	1.9 x 10 <sup>-8</sup> Btu	0.02 mJ	
In Oxygen	6.6 x 10 <sup>-9</sup> Btu	0.007 mJ	5.87

Parameter	$\mathbf{H}_{2}$	$CH_4$	C <sub>3</sub> H <sub>8</sub>
Flammability limits in air (%)	4.0 - 74.2	5.0 - 15.0	2.1 – 9.4
Ignition temperature in air (K)	~855	~925	~770
Ignition energy @ STP (J/cc)	~0.74	~0.97	~0.76
Stoichiometric flame temperature in air (K)	~2580	~2340	~2390
Heat of combustion (kJ g <sup>-1</sup> )	135.4	52.8	40.3
Liquid heat of combustion (MJ per liter)	9.59	22.29	23.56
Gas heat of combustion (MJ m <sup>-3</sup> @ STP)	12.09	37.70	93.48
Temp when gas is heavier than RT air (K)	~21	~162	~444

### Source: NASA, Glenn Research Center Safety Manual

5

# **Cryogenic Systems for MTA Solenoid Magnet**



# **Jryogenic Department**





5

## **Tevatron-style Satellite Refrigerator**



## Installation of 300 KW Screw Compressors





Oil vapor contamination from helium gas is further removed through the outdoors purification skid

Coalescers



# **Refrigeration Modes**





# The MTA Process and Control System

### Functions and Mandates:

- Monitor and control operations Reliability
- ✓ Activate alarms for out-of-limits and hazardous conditions Reliability
- ✓ Collect data for performance studies (complementary to users DAQ) Accuracy

### Safety equipment:

- Flammable gas detectors
- -ODH detectors
- FIRUS system
- Building ventilation
- Flow switch

- Audio and visual alarms
- Crash buttons
- Annunciators
- Interlocks







# Siemens-Moore QUADLOG<sup>™</sup>

→ Combines the beneficial features of a PLC (modularity, ladder logic and sequential programming, high-speed logic solving, and industrial strength) with high safety, high availability, and extensive diagnostics

→ QUADLOG<sup>TM</sup> incorporates continuous PID control, analog I/O and a variety of operator interface options not typically available from a PLC

### Integrated safety features for a fail-safe PLC

- Fault tolerance
- Self-testing software

 Decreased start-up time and minimized downtime with online diagnostics and detailed error reporting

 Easier integration with other control systems via open communications

Monitoring and automated response to pre-defined scenarios







# **PLC Modules**



### **SAM: Standard Analog Module**

Pressure Transducers Heater Electro-valve

### **SDM: Standard Discrete Module**

EV, Pump, ODH, FIRUS  $LH_2$  absorber liquid level too low  $H_2$  presence in the absorber status Purge cabinet status Audio system status Emergency answer status (SD progress) FIRUS (ODH and  $H_2$ ) Interlocks status

### **CDM: Critical Discrete Module**

Oxygen Deficiency Hazard (x4) Flammable gas detectors (x3) He cool-down start status System stop status System reset status

### **VIM: Voltage Input Module**

Level, Pressure, Flow, Temp. (Pt-co)

### **RTM: Resistance Temperature Module**

Platinum and carbon temperature sensors

# **Cryogenic Department**









# **Modified Tevatron Cryogenic Control**



![](_page_13_Picture_0.jpeg)

# Mandates from Fermilab and Engineering Solutions

- 1) Need QUADLOG® Safety PLC for monitoring and automated response to predefined scenarios (hydrogen leak, vacuum rise, person entering interlocked room)
- 2) Establish a large set of procedural controls and written operating procedures
- 3) Meet NEC standards for Class I, Div 2, Group B, (Class I = hazard; Div 2 = hazard sometimes present;Group B = hydrogen)
- Establish nitrogen purge box to meet NEC standards; contained ignition sources not meeting code
- 5) Build intrinsically safe barriers
- 6) All cabling using MC type or PLTC cable
- Used H<sub>2</sub> gas detectors located in the experimental hall and in the gas manifold room for automated response
- 8) Use Excess flow valve on hydrogen gas fill line

9) Provide 'secondary containment' and use buffer tank on vacuum volume10)Limit the MAWP of hydrogen vessel to 0.16 MPa

**Note:** Equipment @ Refrigerator: conventional Tevatron devices → DirectLogic PLC

![](_page_13_Picture_13.jpeg)

![](_page_14_Picture_0.jpeg)

5

# Application 1: KEK convection-type LH<sub>2</sub> Absorber Process and Instrumentation

![](_page_14_Figure_2.jpeg)

![](_page_15_Picture_0.jpeg)

# Application 1: KEK convection-type LH<sub>2</sub> Absorber Process and Instrumentation

![](_page_15_Figure_2.jpeg)

# \*

# Application 1: KEK convection-type LH<sub>2</sub> Absorber Installation and Results

![](_page_16_Figure_2.jpeg)

![](_page_16_Picture_3.jpeg)

![](_page_17_Picture_0.jpeg)

# Application 1: KEK convection-type LH<sub>2</sub> Absorber Installation and Results

An upgrade would be needed to better quantify LH2 absorber performance with longer term tests under stable cooling conditions

- More heat deposed
- → Shorten transfer line to optimize helium usage/reduce heat load
- HTC improperly measured

→Immersed helium temperature sensors; remove warm helium heater and use electrical heater (KEK design: sheathed cartridge inserted in finned aluminum exchanger; temperature monitoring for auto-ignition concerns)

- Improve instrumentation
- $\rightarrow$  Cernox thermometry, use liquid level probe for LH<sub>2</sub> bath

Ref: CEC'05 paper - A. Bross et. al, "An upgrade for the MuCool Test Area", submitted to Cryogenic Engineering Conference 2005, CEC'05

![](_page_17_Picture_11.jpeg)

![](_page_18_Picture_0.jpeg)

# Application 2: Superconducting Solenoid Magnet Process and Instrumentation

![](_page_18_Figure_2.jpeg)

![](_page_19_Picture_0.jpeg)

# Application 2: Superconducting Solenoid Magnet Installation and commissioning

![](_page_19_Picture_2.jpeg)

System was moved from another Fermilab area

- + New PLC for LHe and LN<sub>2</sub> transfers
- + Modified cool-down procedure
- + Internet Rack Monitor (IRM) is used as a gateway from PLC to ACNET

![](_page_19_Picture_7.jpeg)

![](_page_19_Figure_8.jpeg)

![](_page_20_Picture_0.jpeg)

# **Application 3: Forced-flow LH**<sub>2</sub> absorber Process and Instrumentation Proposal

![](_page_20_Figure_2.jpeg)

### Safety issue:

25 liters of LH<sub>2</sub> released into the air and ignited with only a 10 % yield the energy equivalent to 4 kg of TNT

![](_page_20_Picture_5.jpeg)

	Total heat load estimation				
Heat load	(W)	80 K	17 K		
Mechanical Sup	oports	67	6		
Superinsulation		1.5	0.2		
Cryostat windo	NS	-	17		
LH2 pump		-			
Total		68.5	73.2		

### vs. 500 W for total refrigeration system

![](_page_20_Figure_8.jpeg)

# \*

# Application 3: Forced-flow LH<sub>2</sub> Absorber 3-D Conceptual Design

![](_page_21_Picture_2.jpeg)

### LH<sub>2</sub> loop and vacuum vessel

- -Structural study
- -Thermal study
- Hydraulic study (in collaboration with Oxford Univ.)

### -Safety Review Documents

### **Reference papers:**

C. Darve et al., "The Liquid Hydrogen System for the MuCool Test Area", CEC'03, 2003

C. Darve et al., "Cryogenic Design for a Liquid Hydrogen Absorber System", ICEC19, 2002

![](_page_21_Picture_11.jpeg)

![](_page_22_Picture_0.jpeg)

•

# **Concluding comments**

On-going cryogenic efforts at MTA

Hydrogen facility was successfully designed, built and commissioned at Fermilab

> Helium facility (compressor, refrigerator and transfer line) is under fabrication to complete the final infrastructure

Temporary infrastructure has permitted to test the LH<sub>2</sub> convection absorber in summer 2004

> SC solenoid magnet cooling system is installed

 $\succ$  Forced-flow LH<sub>2</sub> absorber design must be completed

MTA "deliverables" shall prove the feasibility of ionization cooling but also the practicality of cryogenic cooling in hazardous environment

→MTA = Fermilab facility fully equipped with cryogenic capacity (Hydrogen and Helium)